TO : Metal Forming, Inc.
100 International Drive Peachtree
City, Georgia 30269
Attn.: Mr. Frank Callis
FROM : Thomas M, Shingler, P.E.
Registered Structural Engineer
President
Design Dynamics, Inc.
DATE : December 8, 2005
SUBJECT : 1-1/2" SNAP LOCK @ 16" wide x 0.040" Aluminum ASTM E-1592
Procedure for the Determination of Wind Uplift Capacity @ $2.0 \mathrm{ft} \& 5.0 \mathrm{ft}$ o/c clips

TEST PROCEDURE : ASTM E-1592
STANDARD TEST METHOD FOB THE STRUCTURAL PERFORMANCE OF SHEET METAL ROOF AND SIDING SYSTEMS BY UNIFORM STATIC AIR PRESSURE DIFFERENCE

PURPOSE : To determine both the Ultimate and Allowable Wind Uplift Capacity of the submitted metal roof system when fastened at various panel clip spacing's.

TEST DATE : November 11, 2005 (4 spans f $5.0 \mathrm{ft} \mathrm{o/c)}$

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\text { (4 spans } 12.0 \mathrm{ft} \text { o/e) }
$$

TEST SPECIMEN : Metal Forming, Inc. 1 1/2" deep SNAP LOCK @ 16 " wide $\times 0.040$ " Aluminum
TEST CHAMBER Composed of a stationary floor mounted steel channel frame capped with a secondary steel channel frame with simulated purlins. The "purlins" attached to a rail on the secondary steel channel frame and consisted of $3^{\prime \prime} \times 3^{\prime \prime} \times 3 / 16^{\prime \prime}$ structural steel tubes in combination with slip-fit/sheaf-screwed 16 gauge steel brake-formed channels representing a light gauge steel purlin flange face.
The composite combination of structural steel tubes and 16 gauge steel channels was designed to behave as purlin supports for the test panel specimen,

TEST CHAMBER : (continued)
The test pressures were applied to the specimen via a perimetertaped 4 mil plastic film (Visqueen) designed to accurately configure to the profile of the test panels. The loose lay-up of the Visqueen film allowed the panel shape and the sidejoint elements a $\cdot$ full degree-of-freedom relative to profile distortion and sidejoint rotation during the loading process.

PURLIN MEMBERS
$: 16$ gauge steel brake-formed channel-sections fit over and were
shear-screwed to the sidewalls of the $3^{\prime \prime} \times 3^{\prime \prime} \times 3 / 16 "$ structural-
tubes. These assemblies were precisely placed at 5.0 ft or
2.0 ft centers to represent the maximum and minimum purlin
spacing extremes for the roof panel being tested.

PRESSURE INDICATOR : Two (2) pressure sensors were attached
to the test chamber and consisted of the following......

One (1) Digital Pressure Indicator manufactured by Micro-Pneumatic Logic, Inc. Model MPL-3200 with a full-range accuracy of $1 / 10^{\prime \prime} \mathrm{H} 20$ and traceable to the National Bureau of Standards

One (1) Alta-Robbins, Inc. U-Tube Water Manometer Model 100 SS with, a 24" scale,

DIGIMATIC CALIPEE : Mitutoyo Series Model No. CD-6" B with a full-range accuracy of 0.001
in.
installation : The secondary frame was inverted and panels were installed with one (1) piece 18 gauge steel hook-style clips which were screw-' attached to the top flanges of the 16 gauge steel channel sections using two (2)
10-16 x 1 " long Pancake Head self-drilling screws with a \#2 Phillips drive slot and \#3 drill point.
Side joints consisted of mating snaptogether male/female interlocking rib elements engaged with hand pressure. Continuity fasteners were located at the extreme ends of panels and consisted of single self-drilling screws driven through the panel webs.

INSTALLATION : (continued)
Both ends of the test assembly were fastened to simulate "free" end conditions. A transparentfflexible plastic film (Visqueen @ 4 mils thick) was loosely applied (pleated) over panels and the secondary steel channel frame was rotated 180 degrees about the longitudinal axis and set down on the stationary floor mounted steel channel frame.

The edges of the Visqueen extended beyond the test assembly limits and were pinched airtight by the dead weight of the secondary steel channel frame.

The individual panels were installed into the test chamber as a five (5) panel wide array per standard field techniques. Specific deflection
PROCEDURE measurement targets were established on key" panels.
These targets and their* locations are illustrated on an enclosed sketch.
Two (2) identical pressure gages were installed onto the test chamber for cross checking of test pressures and insuring accuracy of pressure data.

Initially the system was preloaded to (-) 5 psf to insure proper seating of the panels, clips and plastic film.
After the preloading process, initial deflection measurements were taken at the four (4) key panel locations. These initial deflection readings represented the zero position/zero load specimen status from which all readings were referenced. Individual data sheets and graphic plots of the deflection readings are enclosed with this report.
A "step loading" procedure was used with pressure increments as indicated on enclosed data sheets.

At each incremental pressure level, the test pressure was maintained for an excess of sixty (60) seconds.

After the sixty (80) second pressure "hold" period, measurements were recorded at each of the four (4) critical panel locations.

Also during this time period...broad-flat areas of the panels, sidejoints and clips were visually checked for signs of localized distress.

At the end of each pressure "hold" phase, the test chamber was returned to a zero pressure status and deflection measurements were once again recorded to check for meaningful "set" in the system.
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Ever-increasing pressure values were applied and deflection values recorded both at the pressure as well as at zero.

This procedure continued until the Ultimate Uplift Pressure of the panel or a panel system component demonstrated "distress".

The Allowable Uplift Capacity for the tested panel system is the Ultimate Uplift Pressure divided by a Factor-of-Safety of 1.65. Please note that this Factor-of-Safety value can vary by Project Specification and/or Design Professional.

The Allowable Uplift Pressure for the panel system was established at $2.0 \mathrm{ft} \& 5.0 \mathrm{ft}$, with intermediate Allowable Uplift Pressures being determined via linear interpolation between the two (2) test-established extremes.

## E-1592 TEST RESULTS

| Span, ft | Ultimate Pressure, psf | Allowable Pressure, psf |
| :--- | :--- | :--- |
| 2.0 | 62.4 | 37.8 |
| 5.0 | 33.8 | 20.4 |

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ALLOWABLE WIND UPLIFT LOAD/SPAN CHART :

| PanelS Span, <br> feet | Allowable Wind Uplift <br> Pressure, psf |
| :--- | :--- |
| 2.0 | 37.8 |
| 2.5 | 34.9 |
| 3.0 | 32.0 |
| 3.5 | 29.1 |
| 4.0 | 26.2 |
| 4.5 | 23.3 |
| 5.0 | 20.4 |

General Notes:

1. The Allowable Pressure is the Ultimate Test Pressure divided by a Factor-of-Safety (Load Factor) of 1.65 Note that the Factor-of-Safety value can vary by Project Specification and/or Design Professional.
2. The published Allowable Wind Uplift Pressure considers panel buckling strength, sidejoint disengagement resistance and dip/sidejoint interactive strength only.
3. The clip-to-substrate fastener capacity must be investigated by a design professional and consider the clip pry coefficient where applicable.
4. Tests were conducted by Design Dynamics, Inc. in strict accordance with the ASTM $£-1592$ procedure.
******MD OF REPORT ${ }^{* * * * * * * ~}$
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ASTME-1592 TESTING
SNAP LOCK1 1/2"X195/8" . 040 ALUMINUM

| Pressure in. <br> H20 | Pressure psf | Deflection (in.) |  |  |  |
| :--- | :--- | :--- | :--- | :--- | :--- |
|  |  | 1 | 2 | 3. | 4 |
| 0.000 |  | 2.513 | 2.470 | 2.854 | 2.765 |
| 0.500 |  | 0.002 | 0.03 | 0.26 | 0.251 |
| 0.000 |  | -0.004 | 0.028 | 0.009 | 0.006 |
| 1.000 | 5.200 | 0.020 | 0.062 | 0.429 | 0.439 |
| 0.000 | 0.000 | 0.004 | 0.014 | 0.016 | 0.007 |
| 1.500 | 7.800 | 0.035 | 0.099 | 0.603 | 0.625 |
| 0.000 | 0.000 | 0.006 | 0.015 | 0.033 | 0.024 |
| 2.000 | 10.400 | 0.038 | 0.107 | 0.714 | 0.734 |
| 0.000 | 0.000 | 0.005 | 0.031 | 0.034 | 0.021 |
| 2.500 | 13.000 | 0.048 | 0.133 | 0.833 | 0.85 |
| 0.000 | 0.000 | 0.009 | 0.016 | 0.032 | 0.018 |
| 3.000 | 15.600 | 0.057 | 0.171 | 0.958 | 0.982 |
| 0.000 | 0.000 | 0.005 | 0.022 | 0.031 | 0.019 |
| 3.500 | 18.200 | 0.069 | 0.205 | 1.043 | 1.069 |
| 0.000 | 0.000 | 0.010 | 0.421 | 0.036 | 0.015 |
| 4.000 | 20.800 | 0.081 | 0.234 | 1.116 | 1.148 |
| 0.000 | 0.000 | 0.011 | 0.029 | 0.036 | 0.026 |
| 4.500 | 23.400 | 0.092 | 0.266 | 1.225 | 1.258 |
| 0.000 | 0.000 | 0.019 | 0.029 | 0.039 | 0.041 |
| 5.000 | 26.000 | 0.115 | 0.308 | 1.398 | 1.433 |
| 0.000 | 0.000 | 0.023 | 0.041 | 0.039 | 0.047 |


| 5.500 | 28.600 | 0.145 | 0.346 | 1.535 | 1.561 |
| :--- | :--- | :--- | :--- | :--- | :--- |
| 0.000 | 0.000 | 0.029 | 0.054 | 0.052 | 0.067 |
| 6.000 | 31.200 | 0.188 | 0.421 | 1.76 | 1.79 |
| 0.000 | 0.000 | 0.042 |  |  |  |
| 6.500 | 33.800 | 0.241 | 0.553 | 0.058 | 0.095 |
| 0.000 | 0.000 | 0.060 | 0.115 | 0.126 | 2.123 |

Purlin Spacing: 5'-0"
Node 1: At rib, at purlin
Node 2: At rib, midspan between purlins
Node 3: At panel center, at purlin
Node 4: At panel center, between purlins
Center panel of chamber monitored

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Chamber consisted of full width panels - 5 ribs
Mode of failure: Joint disengagement for full length of panel.

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| Pressure in. H2O | Pressure psf | Deflection (in.) |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | 1 | 2 | 3 | 4 |
| 0.000 | 0.000 | 2.371 | 2.354 | 2.694 | 2.671 |
| 0.500 | 2.600 | -0.010 | -0.005 | 0.173 | 0.183 |
| 0.000 | 0.000 | -0.006 | -0.004 | 0.003 | 1.002 |
| 1.000 | 5.200 | 0.001 | 0.000 | 0.306 | 0.321 |
| 0.000 | 0.000 | -0.006 | -0.010 | 0.005 | 0.001 |
| 1.500 | 7.800 | 0.013 | 0.004 | 0.449 | 0.468 |
| 0.000 | 0.000 | -0.003 | -0.008 | 0.000 | -0.003 |
| 2.000 | 10.400 | 0.019 | 0.013 | 0.559 | 0.583 |
| 0.000 | 0.000 | 0.001 | -0,004 | -0.002 | 0.002 |
| 2.500 | 13.000 | 0.032 | 0.035 | 0.666 | 0.692 |
| 0.000 | 0.000 | 0.012 | 0.020 | -0.002 | 0.014 |
| 3,000 | 15,600 | 0.049 | 0.043 | 0.763 | 0.790 |
| 0.000 | 0.000 | 0.016 | 0.015 | 0.007 | 0.021 |
| 3.500 | 18.200 | 0.044 | 0.048 | 0.836 | 0.865 |
| 0.000 | 0.000 | 0.013 | 0.016 | 0.000 | 0,018 |
| 4.000 | 20.800 | 0.050 | 0.055 | 0.904 | 0,932 |
| 0.000 | 0.000 | 0.018 | 0.016 | 0,003 | 0.019 |
| 4.500 | 23.400 | 0.059 | 0.062 | 0,969 | 0.996 |
| 0.000 | Q.OQO | 0.013 | 0.017 | 0.008 | 0.017 |
| 5.0QO | 26.000 | 0.066 | 0,076 | 1,035 | 1.068 |
| 0.000 | 0.000 | 0.013 | 0.020 | 0.008 | 0.025 |
| 5.500 | 28,600 | 0.064 | 0.076 | 1.109 | 1.142 |
| 0.000 | 0,000 | 0.012 | 0,021 | 0,026 | 0.048 |
| 6.000 | 31.200 | 0.071 | 0.084 | 1.198 | 1,233 |
| 0.000 | 0.000 | 0.016 | 6.027 | 0053. | 0.076 |
| 6.500 | 33.800 | 0.080 | 0.103 | 1,279 | 1,323 |
| 0.000 | 0,000 | 0.467 | 0,020 | 0.073. | 0.1iQ2 |
| 7,000 | 3i:,400 | 0.OB3 | 0.104 | 1.373: | 1.430 |
| 0.000 | 0,000 | O.Q27 | 0.025 | 0.091 | 0,118 |
| 7.500 | 39,000 | 0.094 | 0.110 | 1.482 | 1.531 |
| 0.000 | 0.000 | 0.010 | 0.019 | 0.113 | 0.161 |
| 8.000 | 41.600 | 0.098 | 0.130 | 1.595 | 1.65:8 |
| 0.000 | 0.000 | 0.021 | 0.021 | 0.138 | 0.174 |
| 8.500 | 44.200 | . 0.110 | 0.151 | 1.701 | 1.781 |
| 0.000 | 0,000 | 0,021 | 0.02:5 | 0.151 | 0.203 |
| 9.000 | 46.800 | 0.125 | 0.184 | 1.814 | 1,885 |
| 0.000 | 0.000 | 0.025 | 0,040 | 0,172 | 0.206 |
| 9.500 | 49.400 | 0.127 | Q. 177 | 1.900 | 1.964 |
| 0.000 | 0.000 | 0,021 | ' 0.029 | 0.177 | 0.222 |
| 10.000 | 52.000 | 0.136 | 0,18:8 | 1,970 | 2,044 |
| 0.000 | 0.000 | 0,019 | 0,029 | 0.152 | 0.221 |
| 10.500 | 54.600 | 0.147 | 0,187 | 2.047 | 2.117 |
| 0.000 | 0.000 | 0,021 | 0.039 | 0,0§0 | 0.196 |
| 11.000 | 57.200 | 0.157 | 0.206 | 2.144 | 2.212 |
| 0.000 | 0.000 | 0,018 | 0.032 | 0.117 | 0.204 |
| 11.500 | 59.800 | 0.173 | 0.215 | 2.244 | 2.313 |
| 0.000 | $0 . Q 00$ | 0.034 | 0.036 | 0,126 | 0,215 |
| 12.000 | 62.400 | 0.199 | 0.249 | 2.368 | 2.470 |
| 0.000 | 0.000 | 0.054 | 0,074 | 1.025 | 1,278 |







| Pressure in. <br> H20 | Pressure psf | Deflection (in.) |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | 1 | 2 | 3 | 4 |
| 0.000 | 0.000 | 2.394 | 2.355 | 2.448 | 2.358 |
| 0.500 | 2.600 | -0.002 | 0.024 | 0.018 | 0.117 |
| 0.000 | 0.000 | 0.013 | 0.026 | 0.015 | 0.019 |
| 1.000 | 5.200 | 0.062 | 0.073 | 0.043 | 0.224 |
| 0.000 | 0.000 | 0.034 | 0.041 | 0.018 | 0.016 |
| 1.500 | 7.800 | 0.063 | 0.084 | 0.182 | 0.266 |
| 0.000 | 0.000 | 0.041 | 0.048 | 0.027 | 0.032 |
| 2.000 | 10.400 | 0.079 | 0.093 | 0.192 | 0.365 |
| 0.000 | 0.000 | 0.046 | 0.050 | 0.035 | 0.052 |
| 2.500 | 13.000 | 0.096 | 0.099 | 0.200 | 0.449 |
| 0.000 | 0.000 | 0.029 | 0.049 | 0.024 | 0.032 |
| 3.000 | 15.600 | 0.111 | 0.124 | 1.279 | 1.401 |
| 0.000 | 0.000 | 0.036 | 0.038 | 0.028 | 0.027 |
| 3.500 | 18.200 | 0.109 | 0.131 | 1,452 | 1,541 |
| 0.000 | 0.000 | 0.045 | 0.044 | 0.035 | 0,024 |
| 4,000 | 20.800 | 0.129 | 0.141 | 1.536 | 1.621 |
| 0.000 | 0.000 | 0.027 | 0.045 | 0,032 | 0.041 |
| 4.500 | 23.400 | 0.125 | 0.153 | 1.597 | 1.690 |
| 0.000 | 0,000 | 0.005 | 0.025 | 0,199 | 0.355 |
| 5.000 | 26.000 | 0.119 | 0.156 | 1.891 | 1.790 |
| 0.000 | 0.000 | 0.030 | 0.027 | 0.206 | 0.372 |
| 5.500 | 28.600 | 0.134 | 0.158 | 1.802 | 1.913 |
| 0.000 | 0.000 | 0.030 | 0.023 | 0,210 | 0.374 |
| 6.000 | 31.200 | 0.153 | 0.163 | 1.860 | 1.975 |
| 0.000 | 0.0Q0 | 0.020 | 0,018 | 0.211 | 0,384 |
| 6.500 | 33.800 | 0.150 | 0.170 | 1.996 | 2.113 |
| 0.000 | 0.000 | 0.000 | 0,013 | 0.209 | 0,375 |
| 7.000 | 36.400 | 0.160 | 0,192 | 2.087 | 2.207 |
| 0.000 | 0.000 | 0,017 | 0,005 | 0,200 | 0,371 |
| 7.500 | 39.000 | 0,173 | 0.201 | 2.179 | 2.302 |
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